

**Homework Set 4**  
**Physics 319**  
**Classical Mechanics**

Problem 7.8

a) The Lagrangian is

$$\mathcal{L}(\dot{x}_1, \dot{x}_2, x_1, x_2) = T - U = \frac{m}{2} \dot{x}_1^2 + \frac{m}{2} \dot{x}_2^2 - \frac{k}{2} (x_1 - x_2 - l)^2$$

b) In terms of the center of mass coordinates and  $x$

$$x_{cm} = \frac{mx_1 + mx_2}{2m} = \frac{x_1 + x_2}{2}$$

$$x_1 = x_{cm} + \frac{x_1 - x_2}{2} \quad x_2 = x_{cm} - \frac{x_1 - x_2}{2}$$

$$\dot{x}_1 = \dot{x}_{cm} + \frac{\dot{x}_1 - \dot{x}_2}{2} \quad \dot{x}_2 = \dot{x}_{cm} - \frac{\dot{x}_1 - \dot{x}_2}{2}$$

$$\mathcal{L} = m\dot{x}_{cm}^2 + m \frac{(\dot{x})^2}{4} - \frac{k}{2} (x - l)^2$$

$$\frac{d}{dt}(2m\dot{x}_{cm}) = 0$$

$$\frac{d}{dt}(m\dot{x}/2) = k(x - l)$$

c) The center of mass moves with uniform velocity. The equation of motion for  $x$  is

$$\ddot{x} = \frac{2k}{m}(x - l)$$

$$x(t) = l + A \cos(\omega t + \delta)$$

$$\omega = \sqrt{\frac{2k}{m}}$$

Problem 7.23

If one defines  $x_c = X + x$ , as the inside cart's location in space, the Lagrangian is either

$$\mathcal{L} = T - U = \frac{m}{2} \dot{x}_c^2 - \frac{k(X - x_c)^2}{2}$$

$$\therefore \frac{d}{dt}(m\dot{x}_c) = k(X - x_c)$$

or

$$\mathcal{L} = T - U = \frac{m}{2}(\dot{x} - A\omega \sin \omega t)^2 - \frac{k(x)^2}{2}$$

$$\therefore \frac{d}{dt}(m\dot{x} - mA\omega \sin \omega t) = kx$$

In either case the equation of motion is

$$\ddot{x} + \omega_0^2 x = \omega^2 A \cos \omega t.$$

where  $\omega_0^2 = k/m$ . There is no energy integral for this problem.

### Problem 7.29

It is clear that the point  $P$  is at the location

$$\vec{x}_p(t) = R\hat{x} \cos \omega t + R\hat{y} \sin \omega t.$$

In terms of the given angle the mass is at the point

$$\vec{x}(t) = \vec{x}_p(t) + l \sin \phi \hat{x} - l \cos \phi \hat{y}$$

$$\dot{x}(t) = -R\omega \sin \omega t + l\dot{\phi} \cos \phi$$

$$\dot{y}(t) = R\omega \cos \omega t + l\dot{\phi} \sin \phi$$

$$U = mgy = R \sin \omega t - l \cos \phi$$

The Lagrangian is

$$\mathcal{L} = T - U = \frac{m}{2}(R^2\omega^2 + 2R\omega l\dot{\phi} \sin(\phi - \omega t) + l^2\dot{\phi}^2) - mg(R \sin \omega t - l \cos \phi)$$

The equation of motion is

$$\frac{d}{dt}(ml^2\dot{\phi} + R\omega l \sin(\phi - \omega t)) = mR\omega l\dot{\phi} \cos(\phi - \omega t) - mgl \sin \phi$$

$$l\ddot{\phi} = (R\omega^2 - R\omega\dot{\phi} + R\omega\dot{\phi})\cos(\phi - \omega t) - g \sin \phi$$

$$l\ddot{\phi} = R\omega^2 \cos(\phi - \omega t) - g \sin \phi$$

### Problem 7.34

Consider a uniform expansion over the spring's length. On the fixed side of the spring the coil does not move and the free side it moves  $x(t)$ . If  $s$  measures the distance along the unextended and uncompressed spring with  $s=0$  at the fixed end, then the coil at location  $s$  moves by

$$\Delta x_{coil}(s, t) = \frac{s}{L} x(t)$$

when the expansion in the spring is  $x(t)$ . Similarly, the velocity of the coil at location  $s$  is

$$v_{coil} = \frac{\partial}{\partial t} \Delta x_{coil}(s, t) = \frac{s}{L} \dot{x}(t)$$

The total kinetic energy of the coil is

$$T_{coil} = \int dm \frac{v^2}{2} = \frac{M}{L} \int_0^L \frac{s^2 \dot{x}^2}{2L^2} ds = \frac{M}{6} \dot{x}^2,$$

Which needs to be added to the kinetic energy of the car,  $m\dot{x}^2/2$ . The total Lagrangian is

$$\mathcal{L} = \left[ \frac{m}{2} + \frac{M}{6} \right] \dot{x}^2 - \frac{k}{2} x^2.$$

The usual calculation of the equation of motion leads to sinusoidal oscillations with (angular) frequency  $\omega = \sqrt{k/(m+M/3)}$ .

### Problem 7.35

In the frame that rotates with the wire the position is

$$\vec{x}_r(t) = \frac{AB}{2}(1 + \cos \phi) \hat{x}_r + \frac{AB}{2} \sin \phi \hat{y}_r.$$

But the unit vectors rotate as

$$\hat{x}_r(t) = \cos \omega t \hat{x} + \sin \omega t \hat{y}$$

$$\hat{y}_r(t) = -\sin \omega t \hat{x} + \cos \omega t \hat{y}$$

And therefore

$$x(t) = \frac{AB}{2}(1 + \cos \phi) \cos \omega t - \frac{AB}{2} \sin \phi \sin \omega t = \frac{AB}{2} \cos \omega t + \frac{AB}{2} \cos(\omega t + \phi)$$

$$y(t) = \frac{AB}{2}(1 + \cos \phi) \sin \omega t + \frac{AB}{2} \sin \phi \cos \omega t = \frac{AB}{2} \sin \omega t + \frac{AB}{2} \sin(\omega t + \phi)$$

$$\dot{x}(t) = -\frac{AB}{2} \omega \sin \omega t - \frac{AB}{2} (\omega + \dot{\phi}) \sin(\omega t + \phi)$$

$$\dot{y}(t) = \frac{AB}{2} \omega \cos \omega t + \frac{AB}{2} (\omega + \dot{\phi}) \cos(\omega t + \phi)$$

$$\mathcal{L} = T - U = \frac{m}{2} \left[ \frac{AB^2}{4} (\omega + \dot{\phi})^2 + \frac{AB^2}{2} \omega (\omega + \dot{\phi}) \cos \phi + \frac{AB^2}{4} \omega^2 \right] - 0$$

$$\frac{d}{dt} \left[ m \frac{AB^2}{4} (\omega + \dot{\phi} + \omega \cos \phi) \right] = -m \frac{AB^2}{4} \omega (\omega + \dot{\phi}) \sin \phi$$

$$\ddot{\phi} - \omega \dot{\phi} \sin \phi = -\omega (\omega + \dot{\phi}) \sin \phi \rightarrow \ddot{\phi} = -\omega^2 \sin \phi$$

This result is “obviously” correct as the centrifugal force from the rotation gives an effective potential energy of

$m\omega^2 (AB^2 - r^2)/2 = m\omega^2 (AB)^2 (1 - \cos \phi)/4 \approx m\omega^2 (AB/2)^2 \phi^2/2$ , and the kinetic

energy in the rotating frame is  $m(AB/2)^2 \dot{\phi}^2/2$ . The Euler-Lagrange equations in this

rotating frame give  $\ddot{\phi} = -\omega^2 \sin \phi$  exactly, as they should. The (angular) frequency of small oscillations is  $\omega$ .

### Problem 8.12

a) The first part is straight from the lectures where this radius was defined to be  $r_m$ .

$$r_m = \frac{l^2}{Gm_1 m_2 \mu}$$

b) For small deviations from  $r_m$ , the effective potential is, to second order in the deviation,

$$U_{eff} = E_{min} + \frac{d^2 U_{eff}}{dr^2}(r = r_m) \frac{(\Delta r)^2}{2} + \dots$$

$$\frac{d^2 U_{eff}}{dr^2}(r = r_m) = -2 \frac{Gm_1 m_2}{r_m^3} + 3 \frac{l^2}{\mu r_m^4} = \frac{l^2}{\mu r_m^4}$$

and the motion is a simple harmonic motion with (angular) frequency

$\omega = \sqrt{k / \mu} = l / \mu r_m^2$ . Because for small deviations  $a \approx b \approx r_m$ , this is a restatement of Kepler's second law  $T = 2\pi ab\mu / l$ .

### Problem 8.19

For the ellipse  $a = 3300 \text{ km} / 2 + 6400 \text{ km} = 8050 \text{ km}$ . Therefore

$f = 8050 \text{ km} - 6700 \text{ km} = 1350 \text{ km}$ . The eccentricity is  $1350 / 8050 = 0.1677$ . Because

$r_m = (1 - \varepsilon^2)a = .9719 \times 8050 = 7824 \text{ km}$ , the height at 90 degrees is 1424 km.

### Problem 8.29

From the lectures we know that on a circular orbit of radius  $r_m$ , the total energy and

kinetic energy and potential energy are  $-\frac{(Gm_e m_s)^2 \mu}{2l^2}$ ,  $\frac{(Gm_e m_s)^2 \mu}{2l^2}$ , and

$-\frac{(Gm_e m_s)^2 \mu}{l^2}$ , respectively (consistent with the virial theorem!). When the sun's mass

is halved the angular momentum, and hence the kinetic energy is conserved. At this

same instant, the potential energy becomes  $-\frac{Gm_e m_s}{2r_m} = -\frac{(Gm_e m_s)^2 \mu}{2l^2}$  and so the total

energy vanishes. Without further ado we know that the orbit is a parabola out to  $r = \infty$  with minimum radius  $r_m$ . The parabola is tangent to the original orbit at the instant when the mass changed.

Problem 8.30

The equation for the orbit is

$$r(\theta) = \frac{r_m}{1 + \varepsilon \cos \theta}$$

Because  $x = r(\theta) \cos \theta$  and  $r(\theta) = r_m - \varepsilon x$ ,

$$x^2 + y^2 = (r_m - \varepsilon x)^2 = r_m^2 - 2r_m \varepsilon x + \varepsilon^2 x^2$$

$$x^2(1 - \varepsilon^2) + 2r_m \varepsilon x + y^2 = r_m^2$$

For  $\varepsilon = 1$  the orbit is the parabola

$$y = \pm \sqrt{r_m^2 - 2r_m x} \quad x < r_m / 2$$

This is Eq. 8.60 recalling  $r_m$  is Taylor's  $c$ .

For  $\varepsilon > 1$ , by completing the square the orbit is the hyperbola

$$\left(x + \frac{r_m \varepsilon}{1 - \varepsilon^2}\right)^2 (1 - \varepsilon^2) + y^2 = r_m^2 + \frac{r_m^2 \varepsilon^2}{1 - \varepsilon^2}$$

$$\left(x - \frac{r_m \varepsilon}{\varepsilon^2 - 1}\right)^2 \frac{(\varepsilon^2 - 1)^2}{r_m^2} - \frac{\varepsilon^2 - 1}{r_m^2} y^2 = 1$$

so  $\alpha = c / (\varepsilon^2 - 1)$ ,  $\beta = c / \sqrt{\varepsilon^2 - 1}$ , and  $\delta = c\varepsilon / (\varepsilon^2 - 1)$  in the terms defined by 8.61. Also, one may solve for  $y$

$$y^2 = \left(x - \frac{r_m \varepsilon}{\varepsilon^2 - 1}\right)^2 (\varepsilon^2 - 1) - \frac{r_m^2}{\varepsilon^2 - 1}$$

$$y = \pm \sqrt{\left(x - \frac{r_m \varepsilon}{\varepsilon^2 - 1}\right)^2 (\varepsilon^2 - 1) - \frac{r_m^2}{\varepsilon^2 - 1}} \quad x < \frac{r_m (\varepsilon - 1)}{\varepsilon^2 - 1}$$

This result explains the diagram and equation for the hyperbola in Lecture 15.

Problem 8.33

After the satellite is placed into the transfer orbit the relations

$$R_1 = \frac{c_2}{1 + \varepsilon_2} \quad R_3 = \frac{c_2}{1 - \varepsilon_2}$$

apply. Eliminating  $c_2$  provides the eccentricity as  $\varepsilon_2 = (R_3 - R_1) / (R_3 + R_1)$ . To achieve a circular orbit at  $P'$ ,  $c_3 = R_3$  must be obtained. Therefore

$$\begin{aligned} \frac{c_3}{c_2} &= \lambda'^2 = \frac{1}{1 - \varepsilon_2} \\ &= \frac{R_3 + R_1}{R_3 + R_1 - R_3 + R_1} = \frac{R_3 + R_1}{2R_1} \\ \lambda' &= \sqrt{\frac{R_3 + R_1}{2R_1}} \end{aligned}$$

This is Taylor's Eqn. 8.73.